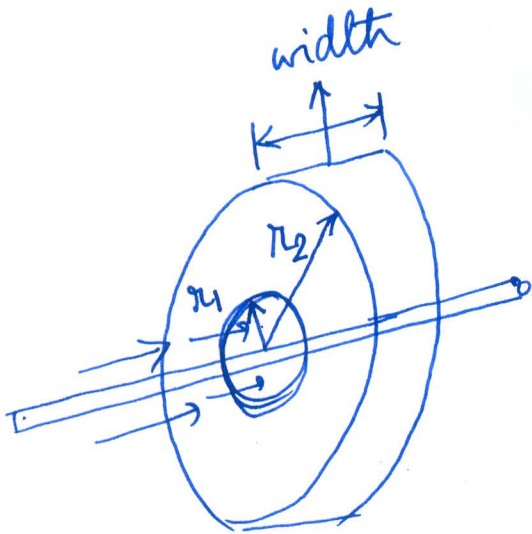
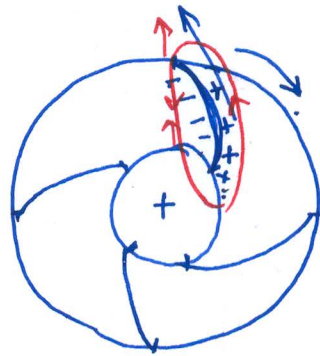


Impeller



runner

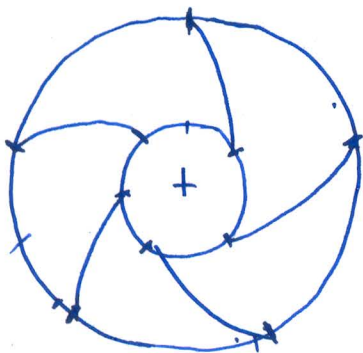
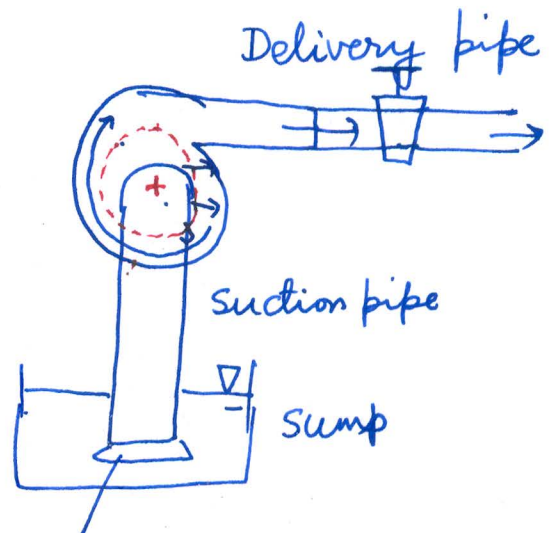
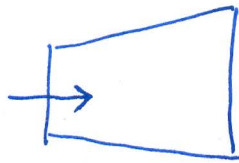


Q m³/s H m of water

$$H = \frac{\rho b}{\rho g}$$

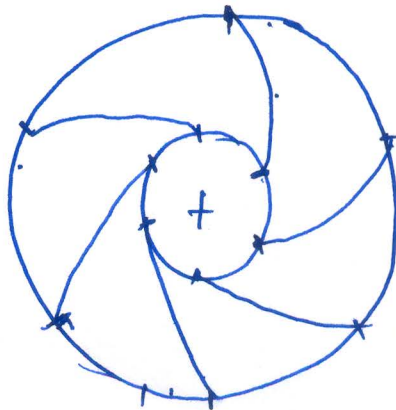
$$p = \rho RT$$

r_1, r_2 length
 N rpm



$$\frac{p}{\rho} + \frac{v^2}{2} + gz = C$$

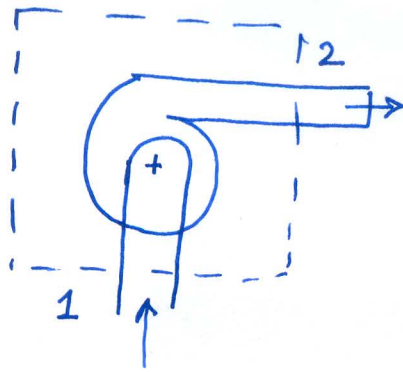
Foot Valve
NRV → Non-return valve



- Direction of flow inside impeller
centrifugal; axial; mixed flow
- energy adding device pump handling incompressible
compressor " α " compressible
- inlet and outlet diameter
 r_1 r_2
- rpm of machine
- direction of rotation of machine
- non-dimensional number decides the type
of machine i.e. centrifugal, axial, mixed flow

Rotodynamic machine \checkmark

Positive displacement machine \times



Steady flow energy equation (SFEE)

$$\dot{Q} - \dot{W} = \dot{m}_2 \left(h + \frac{V^2}{2} + gz \right)_2 - \dot{m}_1 \left(h + \frac{V^2}{2} + gz \right)_1$$

$\dot{Q} = +ve$ energy added Reynolds Transport

$\dot{W} = +ve$ work done Theorem (RTT)

$\left(h + \frac{V^2}{2} + gz \right)_2$ flux out

$\left(h + \frac{V^2}{2} + gz \right)_1$ flux in

$\dot{m}_2 = \dot{m}_1 = \dot{m}$ continuity equation

Turbomachine adiabatic $\dot{Q} = 0$

$$-\dot{W} = \dot{m} \left(h + \frac{V^2}{2} + gz \right)_2 - \dot{m} \left(h + \frac{V^2}{2} + gz \right)_1$$

$$w = \text{specific work} = \frac{\dot{W}}{\dot{m}} = \frac{\text{J/sec}}{\text{kg/sec}} = \text{J/kg}$$

$$-w = \left(h + \frac{V^2}{2} + gz \right)_2 - \left(h + \frac{V^2}{2} + gz \right)_1 \Rightarrow \text{energy absorbing device}$$

$$w = \left(h + \frac{V^2}{2} + gz \right)_1 - \left(h + \frac{V^2}{2} + gz \right)_2 \Rightarrow \text{energy extracting device}$$

$$h = u + pv$$

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(4)

$h =$ enthalpy J/kg

$u =$ internal energy J/kg

$p =$ pressure Pa

$v =$ specific volume m^3/kg

$$-w = \left(u + pv + \frac{v^2}{2} + gz \right)_2 - \left(u + pv + \frac{v^2}{2} + gz \right)_1$$

Incompressible :

$$-w = \left(pv + \frac{v^2}{2} + gz \right)_2 - \left(pv + \frac{v^2}{2} + gz \right)_1 + (u_2 - u_1)$$

$$u_2 - u_1 \approx 0$$

$$cv(T_2 - T_1) \approx 0$$

$$T_2 \approx T_1$$

$$-w = \left(\frac{p}{\rho} + \frac{v^2}{2} + gz \right)_2 - \left(\frac{p}{\rho} + \frac{v^2}{2} + gz \right)_1$$

Turbine

$$w = \left(\frac{p}{\rho} + \frac{v^2}{2} + gz \right)_1 - \left(\frac{p}{\rho} + \frac{v^2}{2} + gz \right)_2$$

\dot{Q} = Volume flow rate
m³/s

H = Head, m

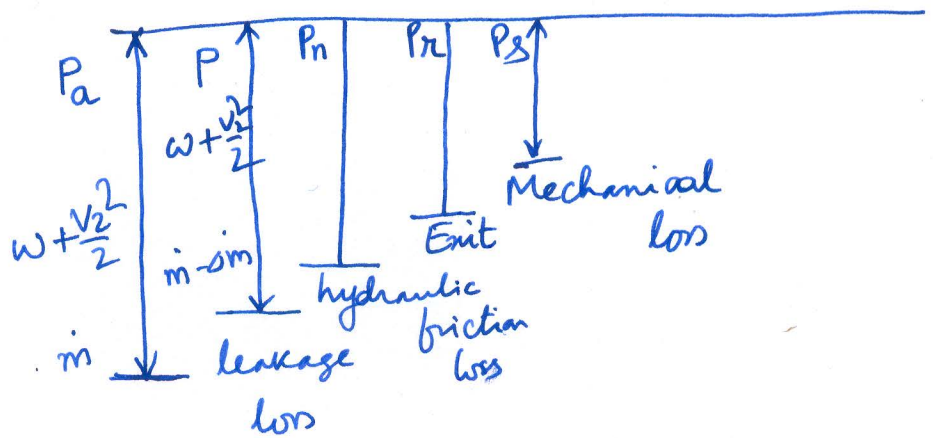
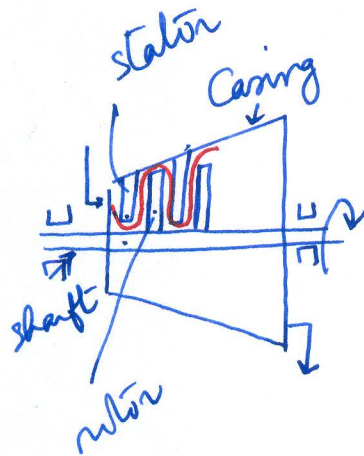
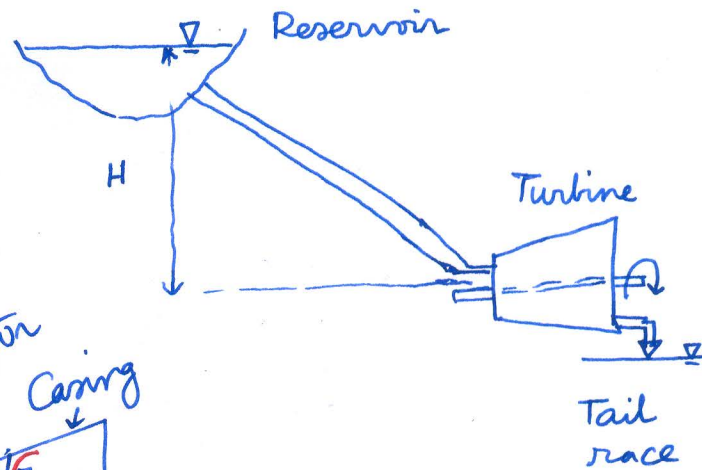
ρ = Density, kg/m³

$$\rho g H$$

$$\dot{m} = \rho \dot{Q} \quad \frac{\text{kg}}{\text{sec}}$$

$$w = \frac{\dot{m} g H}{\dot{m}} = g H$$

$$w + \frac{v^2}{2}$$



Turbomachines

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