4.
$$D_m = 50 \text{ cm}$$
.
 $N = 10000 \text{ spm}$.
 $V_{4} = 380 \text{ m/8}, \text{ s} = 25^{\circ}$
 $\beta_{2} = 35^{\circ}$.

$$R = \frac{\omega - \frac{v_1 - v_2}{2}}{\omega}$$

VICV

to reduce the similarity value.

Turbulent Offset Jet Flows", Numerical Heat Transfer Part A Applications, 1/2008, etc. Please rewrite substantially

· V1 = 60 MB D1 = 90 cm, 01 = 300 N=1200 rpm. D=30 cm. $U_1 = \frac{277N}{60} \times \frac{D_1}{2} = \frac{277 \times 1200}{60} \times \frac{0.9}{2} = 1877 \text{ M/s}$ V2= V12+U2-2U, V, Cosay $=60^{2} \pm (18\pi)$ - 2x60 x 1871 x Cos 30 = 921 m/s = Von = 30,3 m/s Visind= Von Sip, =) simp = 60 x sin30 = 0.99 B1=81.890. teng= Vising = 60 x sin 30 = 1.59 By=57.8°. w= U,Vt1-UzVtz

$$\omega = \frac{V_1^2 - V_2^2}{2(1-R)} = \frac{2938}{2(1-R)} = \frac{V_1^2 - V_2^2}{2(1-R)} = \frac{2938}{2(2938)} = \frac{2938}{2(2938)} = \frac{2938}{2(2938)} = \frac{1-R}{2} = \frac{60^2 - (60 \times 8in 30)^2}{2 \times 2938}$$

$$\begin{aligned}
& = \frac{V_1^2 - V_2 L}{V_1^2 - RV_2 L^2} = \frac{60^2 - 30^2}{60^2 - 0.54} \\
& = \frac{675}{1469} = 0.54 \\
& = \frac{675}{1469} = 0.54 \\
& = \frac{6}{7} = 0.86
\end{aligned}$$



$$\beta_{i} = \frac{100 \times 8 \text{m}^{28}}{88.3 - 36.6}$$

$$\omega = U \left[V_{u_1} - V_{u_2} \right]$$

$$= U [88.3 - (36.6 - 88.3)]$$

88.3)
$$W = \frac{(v_1^2 - v_2^2)}{2} + (v_{22}^2 - v_{2n}^2)$$

$$= \frac{(v_1^2 - v_2^2)}{2} + \frac{(v_{22}^2 - v_{2n}^2)}{2}$$
Symmetry $R = 0.5$

U= 125 m/s B = 40°, B2 = 25° V = 235 m/s. Vr2=1.5 Vr4 Vry cosp3, + V, cos 04 = Vz cosp2 + 1/2 cosx2 Vry sing; = V, tindy Vry Cosp, sing, of V, sind Very tang, Very Cosp, = V, Sinfay, cotp, V, sindy cotps/+V, cosdy =/1 225 x cot 40 x sindy + 2\$5 cos 4 = 125 sind + 275 / Cos 4 = V1= 0+V, -20 Voy COSB, And + 0.84 cos & = 0.47 225 = 125 + x - 2x125x xx Col4 4= \$5. X 35000 = x2+191.5x 22775 22775 D 70.1 > 21 = 18.6°. 109 sinto tan of = 125+109 4840 Vrz=1.5 x 109 = 163.5 m/s. Vuz= Vr Cus /2=163.5 x lux 25-U = -191.5 ± 420.3 = 148.24s-U=23.2 W= U(Vyy-Vyz) = 125x (208.5 +148.2) = 28962.5 J/hg. = 44.4 m/s 109 m/s. V2= 1257 163.52 2x125x163.5 x 6525 220 =125 +x-2x125xxx WS140 V2= 72.9 MS M243 = 200-729 = 21542.8 Ms. x2+191.5x-32775 20 -191.5± /1915+4×32775 R= 28962.5-21542.8 = 0.26 G= 28962.5 28962.5+ 729 = 0.62.92 = -19151 409-6 The Z 200 sin 18.6 - 1635 525 2109 m/s. = 1.07 N(hols)

No whire,
$$\alpha_1 = 90$$

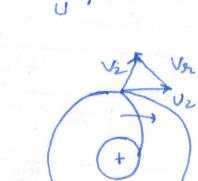
$$V_1 = 12 \text{ m/s}$$

$$U_1 = \frac{2TIN}{60} \times \frac{D_1}{2}$$

$$=\frac{2\Pi\times2000}{60}\times\frac{0.1}{2}$$

$$tan \beta_1 = \frac{V_1}{U} = \frac{12}{10.5}$$

$$=\sqrt{10.5^2+12^2}$$



$$R = \frac{369.9 - (22.3 - 12^2)/2}{369.9}$$